

Research Article

# Permanent Magnet Synchronous Motor Dynamic Modeling with State Observer-based Parameter Estimation for AC Servomotor Drive Application

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#### Abstract

This paper presents model-based control with parameters identifies based on extended Luenberger observer of PMSM drive. To control the permanent magnet synchronous motor for high-performance operation, it is still challenging due to its nonlinear properties and unknown parameters. Therefore, to deal with this issue, nonlinear control with parametric identification can offer high-performance control. Both simulations by MATLAB/ SIMULINK software and the laboratory experiments were carried out in this research. Simulation and experimental results with a small-scale SPMSM of 6-pole, 1-kW, and 3000 rpm in a laboratory corroborate the control scheme capability during a motor-driven cycle.

Keywords: SPMSM, Extended Luenberger Observer (ELO), and Model-based control

Please cite this article as: S. Sriprang, B. Nahid-Mobarakeh, N. Takorabet, S. Pierfederici, N. Bizon, P. Kuman, and P. Thounthong, "Permanent magnet synchronous motor dynamic modeling with state observer-based parameter estimation for AC servomotor drive application," *Applied Science and Engineering Progress*, vol. 12, no. 4, pp. 286–297, Oct.–Dec. 2019.



#### 1 Introduction

Permanent Magnet Synchronous Motor (PMSM) is popularly used in AC servo drives applications more than other motor types due to the significant advantages such as high power ratio, small volume, and simple structure. These applications require both the rapid electrical response and exact knowledge of the uncertainty disturbance torque as well as the system losses to provide a good transient performance for the system. However, rapid dynamic control of a PMSM is not easy owing to the nonlinearities of PMSM, parameters variation, external disturbance torque, unmodeled uncertainties, etc., which can seriously degrade the performance quality of the motor-driven systems. Thus, the linear control schemes such as proportionalintegral control cannot assure required performances. To get around this problem, many researchers have proposed diverse control design methods, e.g., adaptive control [1], neural network control [2], nonlinear feedback linearization control [3], disturbance-observer-based control [4], model predictive control [5], fuzzy-logic-based controller [6], robust control [7] and the combination of these concepts [8]. Recently, several researchers such as [9] and [10] have proposed PMSM control design methods based on the flatness properties of the system called "flatness-based control", which provides a convenient framework for controlling and dealing with nonlinearities or uncertainties. The advantages of flatness-based control are the system's trajectories are straightforwardly estimated by the flat output trajectories and its derivatives without integrating any differential equation.

In this paper, a controller design method using flatness-based control is proposed. And also in order to improve the performance of the control system, machine's parameters including systems losses and external disturbance are going to be estimated by extended Luenberger observer.

In the following sections, a detailed theoretical assay of the proposed method is presented, and simulation results by using MATLAB/Simulink program are provided to prove its effectiveness. Lastly, practical implementation results based on the dSPACE 1104 DSP system are shown to corroborate its correctness.



Figure 1: A three-phase inverter driving the PMSM.

#### 2 Propose Modeling and Controller Design

#### 2.1 Mathematics model of the PMSM/inverter

Figure 1 shows a system configuration of a threephase inverter connected to the PMSM. The sinusoidal pulse-width modulation technique (SPWM) is applied to DC-AC inverter to achieve a sinusoidal output voltage with minimal undesired harmonics. The classic rotating reference frame equations of the non-salient PMSM, which define inductances  $L_s = L_d = L_q$ , are [9]–[12] [Equations (1) and (5)]:

$$\frac{di_d}{dt} = \frac{1}{L_s} (v_d - R_s \cdot i_d - \omega_e \cdot L_s \cdot i_q) \tag{1}$$

$$\frac{di_q}{dt} = \frac{1}{L_s} (v_q - R_s \cdot i_q - \omega_e \cdot L_s \cdot i_d + \Psi_m)$$
(2)

$$\frac{d\omega_m}{dt} = \frac{1}{J} (T_e - B_f \omega_m - T_L)$$
(3)

where

$$T_e = n_p \cdot \Psi_m \cdot i_q \tag{4}$$

$$\omega_e = n_p \cdot \omega_m \tag{5}$$

 $v_d$  and  $v_q$  are the d, q-axis voltages,  $i_d$  and  $i_q$  are the d-, q-axis stator currents,  $R_s$  and  $\Psi_m$  are the resistance (or system losses) and permanent magnet flux linkage, respectively; and  $\omega_e$ ,  $\omega_m$ ,  $n_p$ ,  $T_e$ ,  $T_L$ ,  $B_{fs}$ , J are electrical angular frequency, mechanical angular frequency, number of pole pairs, electromagnetic torque, load torque, viscosity, and inertia, respectively.

### 2.2 Flatness-based control design

Block diagram of all the control algorithm shows in Figure 2 consisted of two main parts, which are the flatness-based control and state observer. Initially, flatness-based control analyzing is described by considering PMSM model and defining  $L_s = L_q = L_d$  for non-salient machine. Flat output candidate y, control variable u, and state variable x are defined that  $y = [y_1 y_2 y_3]^T = [i_d i_q \omega_m]^T$ ,  $u = [u_1 u_2 u_3]^T = [v_d v_q T_e]^T$ , and  $x = [x_1 x_2 x_3]^T = [i_d i_q \omega_m]^T$ , respectively. To prove that PMSM model is said to be differentially flat, the control variable u has to be written as a function of the flat output and their successive derivative [inverse dynamic] that is [Equations (6) and (7)]

$$u_{1} = L_{s} \cdot \frac{di_{d}}{dt} + R_{s} \cdot i_{d} - \omega_{e} \cdot L_{s} \cdot i_{q} = \psi_{1}(y_{1}, \dot{y}_{1}, y_{2})$$
(6)  
$$u_{2} = L_{s} \frac{di_{q}}{dt} + R_{s} \cdot i_{q} + \omega_{e} \cdot (L_{s} \cdot i_{d} + \Psi_{m}) = \psi_{2}(\dot{y}_{2}, y_{2}, y_{1})$$
(7)

By referring to Equation (4), the developed torque equation is proportional to q-axis current that is [Equation (8)]

$$T_e = K_t \cdot i_a \tag{8}$$

where

 $K_t$  is defined as the torque constant that  $K_t = n_p \cdot \Psi_m$ .

Therefore, the  $T_e$  is chosen as a control variable  $u_3 = T_e = K_t \cdot i_q$ . To prove that the system Equation (3) is said to be differentially flat, the input (control) variable  $u_3$  is written as a function of the flat output and its successive derivative (inverse dynamic) that is [Equation (9)]

$$u_{3} = \left(J \cdot \frac{d\omega_{m}}{dt} + T_{L} + B_{f} \cdot \omega_{m}\right) / K_{t} = \psi_{3}(y_{3}, \dot{y}_{3}) = i_{qCOM}$$
(9)

#### 2.3 Control law and stability

If all the parameters of the machine are completely known, flatness-based control won't use any controller, but in practice, those are varied all the time by external disturbance. Therefore, the controller has to be discussed to deal with this. Initially, the controller law can defined as follow [Equation (10)]:

$$\lambda = u_{REF} + u_{feedback}(\varepsilon) \tag{10}$$



**Figure 2**: Structure of the PMSM drive system with the flatness-based control and ELO.



Figure 3: Control law block diagram.

where

 $\varepsilon$  is an error between  $y_{ref}$  and ymeas,  $\varepsilon = y_{ref} - y_{meas}$ .  $\lambda$  is the control law output.

Figure 3 shows block diagram of control law for flatness-based control.

As shown in Figure 3, the intelligent proportionalintegral controller (iPI) [13] is utilized to compensate an error between reference and measured value. By doing that, the two equations of current loop are defined as follows:

$$\lambda_{1} = R_{s} \cdot \varepsilon_{1} + L_{s} \cdot \frac{d\varepsilon_{1}}{dt} + K_{p1} \cdot \varepsilon_{1} + \int K_{i1} \cdot \varepsilon_{1} dt$$
(11)

$$\lambda_2 = R_s \cdot \varepsilon_2 + L_s \cdot \frac{d\varepsilon_2}{dt} + K_{p2} \cdot \varepsilon_2 + \int K_{i2} \cdot \varepsilon_2 dt \qquad (12)$$

where

 $\varepsilon_1$  is an error between  $y_{1REF}$  and  $y_1$ ,  $\varepsilon_1 = y_{1REF} - y_1$ .  $\varepsilon_2$  is an error between  $y_{2REF}$  and  $y_2$ ,  $\varepsilon_2 = y_{2REF} - y_2$ .  $K_{p1}$ ,  $K_{i1}$ ,  $K_{p2}$ , and  $K_{i2}$  are controller parameters.

If the controller is satisfy designed, Equations (11) and (12) will converge to zero that is

$$(R_s + K_{p1}) \cdot \varepsilon_1 + L_s \cdot \frac{d\varepsilon_1}{dt} + \int K_{i1} \cdot \varepsilon_1 dt = 0$$
(13)

$$(R_s + K_{p2}) \cdot \varepsilon_2 + L_s \cdot \frac{d\varepsilon_2}{dt} + \int K_{i2} \cdot \varepsilon_2 dt = 0$$
(14)

Controller parameters can be found out by taking the time-derivative of the Equations (13) and (14) expressed as follows:



$$\ddot{\varepsilon}_{1} + \left(\frac{R_{s} + K_{p1}}{L_{s}}\right) \cdot \dot{\varepsilon}_{1} + K_{i1} \cdot \varepsilon_{1} = 0$$
(15)

$$\ddot{\varepsilon}_2 + \left(\frac{R_s + K_{p2}}{L_s}\right) \cdot \dot{\varepsilon}_2 + K_{i2} \cdot \varepsilon_2 = 0$$
(16)

After that, by comparing Equations (15) and (16) to  $2^{nd}$  order system standard equation, controller parameters will obtain as follows [Equation (17)]:

$$\ddot{q}_1 + 2\zeta \omega_{n1} \dot{q}_1 + \omega_{n1}^2 q_1 = 0$$
(17)

Consequently, controller parameters obtained are [Equations (18) and (19)]

$$K_{p1} = K_{p1} = 2\zeta_1 \omega_{n1} \cdot L_s - R_s$$
(18)

$$K_{i2} = K_{i2} = \omega_{n1}^2 - L_s \tag{19}$$

where

 $\zeta_1$  and  $\omega_1$  are the desired dominant damping ratio and natural frequency, respectively.

Next, the controller of speed loop control is

$$\lambda_3 = \left(J \cdot \frac{d\varepsilon_3}{dt} + T_L + B_j \cdot \varepsilon_3 + K_{p3} \cdot \varepsilon_3 + \int K_{i3} \cdot \varepsilon_3 dt\right) / K_i \quad (20)$$

where

 $\varepsilon_3$  is an error between  $y_{3REF}$  and y3,  $\varepsilon_3 = y_{3REF} - y3$ .  $K_{p3}$ , and  $K_{i3}$  are controller parameters.

If the controller is satisfy designed, Equation (20) is going to converge to zero that is

$$J \cdot \frac{d\varepsilon_3}{dt} + T_L + (B_f + K_{p3}) \cdot \varepsilon_3 + \int K_{i3} \cdot \varepsilon_3 dt = 0$$
(21)

Taking the time-derivative of the Equation (21) yields

$$\ddot{\varepsilon}_3 + \left(\frac{B_f + K_{p3}}{J}\right) \cdot \dot{\varepsilon} + K_{i3} \cdot \varepsilon_3 = 0$$
(22)

Comparing Equation (22) to second-order system standard equation obtains [Equation (23)]

$$\ddot{q}_2 + 2\zeta \omega_{n_2} \cdot \dot{q}_2 + \omega_{n_2}^2 \cdot q_2 = 0$$
(23)

Consequently, controller parameters obtained are [Equations (24) and (25)]

$$K_{p1} = 2\zeta_2 \omega_{n2} \cdot J - B_f \tag{24}$$

$$K_{i2} = \omega_{n2}^2 - J$$
 (25)

where

 $\zeta_2$  and  $\omega_2$  are the desired dominant damping ratio and natural frequency, respectively.

#### 2.4 Trajectory Planning

As shown in Figure 3, the trajectory planning is a crucial part of fatness-based control because it helps to improve the input reference  $y_{REF}$ . The low-pass 2nd filter utilizes to plan the desired trajectory for the flat output component. It allows limiting the derivative terms. Then, planned trajectories of the reference flat output for current and speed loop control are expressed as follows [Equations (26) and (27)]:

$$\frac{y_{1REF}(s)}{y_{1COM}(s)} = \frac{y_{2REF}(s)}{y_{2COM}(s)} \frac{1}{\left(\frac{s}{\omega_{n3}}\right)^2 + \frac{2\zeta_3}{\omega_{n3}}s + 1}$$
(26)

$$\frac{y_{3REF}(s)}{y_{3COM}(s)} = \frac{1}{\left(\frac{s}{\omega_{n4}}\right)^2 + \frac{2\zeta_4}{\omega_{n4}}s + 1}$$
(27)

where  $\zeta_3$ ,  $\zeta_4$ ,  $\omega_{n3}$ , and  $\omega_{n4}$  are the desired dominant damping ratio and natural frequency.

#### 2.5 Parameters and State Estimation [14]–[23]

As depicted in Figure 4, there are losses in terms of resistance and switching devices as well as an external disturbance. Besides, owing to the flatness-based control is model-based control; its performance relies on all the parameters of the system. To enhance the system's performance, they have to be estimated by the use of observer approach. The two observer methods are going to introduce and carry out to observe the parameters and state variable of the system. One of them is a linear observer; the other is a nonlinear observer approach.

Initially, extended Luenberger observer is discussed by considering PMSM model. It is due to the non-salient machine that the developed torque is from only the magnetic alignment torque produced by the flux

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Figure 4: Equivalent of PMSM drive system.

linkage. Therefore, *d*-axis current is set to be zero to reduce copper losses of the machine. Consequently, by neglecting the *d*-axis differential Equation (1) and substituting term  $R_s \cdot i_q$  with  $v_{tq}$  as well as combining the derivative of torque and  $v_{tq}$ , all the system can be rewritten as follows:

$$\frac{di_q}{dt} = \frac{1}{L_s} \left( v_q - v_{tq} - \omega_e \cdot L_s \cdot i_d + \Psi_m \right)$$
(28)

$$\frac{d\omega_m}{dt} = \frac{1}{J} \left( T_e - B_f \omega_m - T_L \right)$$
(29)

$$\frac{dT_L}{dt} = 0 \tag{30}$$

$$\frac{dv_{tq}}{dt} = 0 \tag{31}$$

 $dT_L/dt$  and  $dv_{tq}/dt$  can be considered as zero because their time constant is much larger than for a controller. As it a state observer dedicated to the linear system, it is necessary to linearize the considered system around one operating point. Writing the linearization model of PMSM in state-space form yields [Equation (32)]

$$\mathbf{A} = \begin{bmatrix} 0 & -\frac{\Psi_m}{L_s} & -\frac{1}{L_s} & 0\\ \frac{n_p \cdot \Psi_m}{J} & -\frac{B_f}{J} & 0 & -\frac{1}{J}\\ 0 & 0 & 0 & 0\\ 0 & 0 & 0 & 0 \end{bmatrix}$$



Figure 5: Full-state observer block diagram.

$$\mathbf{B} = \begin{bmatrix} \frac{1}{L_q} & -n_p \cdot \omega_{m0} \\ 0 & 0 \\ 0 & 0 \\ 0 & 0 \end{bmatrix}, \mathbf{C} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix}$$
(32)

where  $x(t) = [i_q \ \omega_m \ v_{tq} \ T_L]^T$ , output variables are  $y = [i_q \ \omega_m]$  and input variables are  $u = [v_q \ i_d]^T$ .

According to Luenberger [24], the full-state observer for the system [Equations (33) and (34)]

$$\dot{x}(t) = \mathbf{A}x(t) + \mathbf{B}u(t) \tag{33}$$

$$y(t) = \mathbf{C}x(t) \tag{34}$$

is given by

$$\dot{\hat{x}} = \mathbf{A}\hat{x} + \mathbf{B}u + \mathbf{L}\left(\mathbf{C}x - \mathbf{C}\hat{x}\right)$$
(35)

where  $\hat{x}$  denotes the estimate of state *x*.

The observer is depicted in Figure 5. The observer has two inputs, u and y, and one output,  $\hat{x}$ . The goal of the observer is to provide an estimate  $\hat{x}$  so that  $\hat{x} \rightarrow x$  as  $t \rightarrow \infty$ . Define the observer estimation error as [Equation (36)]

$$e(t) = x(t) - \hat{x}(t)$$
 (36)

One the main results of the systems theory is that if the system is completely observable, the matrix  $\mathbf{L}$  can be always fined so that the tracking error is asymptotical stable, as desired.

Taking the time-derivative of the estimation error in Equation (37) yields

$$\dot{e} = \dot{x} - \dot{\hat{x}} \tag{37}$$



and using the system model and the observer in Equation (35), it obtains [Equations (38) and (39)]

$$\dot{e} = \mathbf{A}x + \mathbf{B}u - \mathbf{A}\hat{x} - \mathbf{B}u - \mathbf{L}(\mathbf{C}x - \mathbf{C}\hat{x})$$
(38)

or

$$\dot{e}(t) = (\mathbf{A} - \mathbf{L}\mathbf{C}) \cdot e(t) \tag{39}$$

It is evident that  $e(t) \rightarrow 0$  as  $t \rightarrow \infty$  for initial tracking error e(t0) if the characteristic equation

$$\det\left(\lambda \mathbf{I} - (\mathbf{A} - \mathbf{L}\mathbf{C})\right) = 0 \tag{40}$$

has all its roots in the left haft-plane. Therefore, the observer design process reduces to finding the matrix **L** such that the roots of the characteristic equation in Equation (40) lie in the left haft-plan. This con always be accomplished if the system is completely observable; that is, if the observability matrix  $\mathbf{P}_0$  has full rank. [Equation (41)]

$$\mathbf{P}_0 = \begin{bmatrix} \mathbf{C} \\ \mathbf{C}\mathbf{A} \end{bmatrix}$$
(41)

According to Equation (31), the estimated error  $e(t) = \hat{x}(t) - x(t)$  tends to zero if the observer gain L is satisfying designed. Then to achieve this, the eigenvalues of **[A-LC]** was established at  $[(-10000 - 10000 - 18 - 30]^T$ . Those values have been tuned experimentally to obtain the better performances as possible. For the operating point, at speed (n) = 1500 rpm  $\omega_{m0}$  = 157.0796 rad/sec, and  $i_{d(0)} = 0$ , the matrix L is obtained by Equation (42). For this estimation, even if the system has been linearized around one operating point, it has been experimentally verified that the estimation was converging in the speed range 0–1500 rpm with no change of the value of the matrix L. The closed-loop system pole locations can be arbitrarily placed if and only if the system is controllable [Equation (42)].

$$\mathbf{L} = \begin{bmatrix} 260 & -18.811\\ 395.357 & 230.9174\\ -423.720 & 0\\ 0 & -11.088 \end{bmatrix}$$
(42)

Next, a new observer is introduced to compare to the extended Luenberger observer. It can adapt to

the considered problematic consisting of parametric estimation. There was a previous publication that utilized this method to estimate the losses of the boost converter [23]. The proposed state observer is devoted to the subclass of nonlinear systems which can be described as follows:

$$\begin{cases} \dot{\mathbf{X}} = \begin{pmatrix} \dot{x} \\ \dot{d} \end{pmatrix} = \begin{pmatrix} f(x,u) + g(x,u) \cdot d \\ 0 \end{pmatrix} \\ \mathbf{Y} = x \end{cases}$$
(43)

where:

1)  $\dot{\mathbf{X}} \in \mathbb{R}^{n^*m}$  is the vector of the variable which is going to be estimated, and  $Y \in \mathbb{R}^{n^*m}$  is the vector of measured variable;

2)  $x \in \mathbb{R}^n$  is the vector of the system state variable. Every state variable is supposed to be measured (i.e.,  $\mathbf{Y} = x$ );

3)  $d \in \mathbb{R}^m$  is the vector of unknown parameters to estimate. Variable d is supposed to very slowly compared to state variables *x*;

4) *f* and *g* are nonlinear functions of *x* and *u* (the command signal vector), respectively, of size  $\mathbb{R}^n$  and  $\mathbb{R}^{n^*m}$ 

Defining state variable *x*, unknown parameters *d*, *f*, and *g*, respectively that is  $x = [i_q \omega_m]^T$  and  $d = [v_{\iota_q} T_L]^T$ from Equations (28)–(31) into Equation (43) yields [Equations (44) and (45)] [25], [26]

$$f(x,u) = \begin{bmatrix} \frac{1}{L_s} \left( v_q - \omega_e \cdot L_s \cdot i_d - \omega_e \cdot \Psi_m \right) \\ \frac{1}{j} \left( p \cdot \Psi_m \cdot i_q - B_f \cdot \omega_m \right) \end{bmatrix}$$
(44)

$$g(x,u) = \begin{bmatrix} \frac{-1}{L_s} & 0\\ 0 & \frac{-1}{J} \end{bmatrix}$$
(45)

The estimation error of the state and unknown parameters are  $e_x = \hat{x} - x$  and  $e_d = \hat{d} - d$ , state observer definition is defined as follows:

$$\begin{pmatrix} \dot{\hat{x}} \\ \dot{\hat{d}} \\ \dot{\hat{d}} \end{pmatrix} = \begin{pmatrix} f(x,u) + g(x,u) \cdot \hat{d} - \mathbf{S} \cdot e_x \\ \mathbf{K}_p \cdot \dot{e}_x + \mathbf{K}_i \cdot e_x - g^t(x,u) \cdot e_x \end{pmatrix}$$
(46)

with:





$$\mathbf{K}_{p} = -\mathbf{P} \cdot g^{-1}(x, u) \tag{47}$$

$$\mathbf{K}_i = \mathbf{K}_p \cdot \mathbf{S} \tag{48}$$

**S** is the positive-definite matrix of size  $\mathbb{R}^{n^*m}$ . **P** is the positive-definite matrix of size  $\mathbb{R}^{n^*m}$ .

The proposed state observer has been designed in order to obtain a system for which the exponential stability can be proven. Especially, for the part of the system corresponding to the estimation of the parameters p—bottom of Equation (46) the term  $K_p \cdot \dot{e}_x$  and  $K_i \cdot e_x$  have been added so as to verify the exponential stability. Indeed, without those terms (i.e., only considering  $\hat{p} = -g^T(x,u) \cdot e_x$ , only the asymptotic stability can be demonstrated.

For the stability proving of proposed observer, the time-derivative of the estimation error  $e_x$  and  $e_d$ are written, respectively, as follows:

$$\dot{e}_x = \dot{\hat{x}} - \dot{x} = g(x, u) \cdot e_d - \mathbf{S} \cdot e_x \tag{49}$$

$$\dot{e}_{d} = \hat{d} - \dot{d} = \mathbf{K}_{p} \cdot g(x, u) \cdot e_{d} - \mathbf{K}_{p} \cdot \mathbf{S} \cdot e_{x} + \mathbf{K}_{i} \cdot e_{x} - g^{t}(x, u) \cdot e_{x}(50)$$

The exponential stability of the estimation is represented with the classical Lyapunov approach. The candidate function, V is defined by

$$V = \frac{1}{2} \cdot \begin{pmatrix} e_x & e_d \end{pmatrix} \cdot \begin{pmatrix} e_x \\ e_d \end{pmatrix} \ge 0$$
(51)

Taking the time-derivative of the Equation (51) yields

$$\dot{V} = e_x^t \cdot \dot{e}_x + e_d^t \cdot \dot{e}_d \tag{52}$$

By substituting Equations (49) and (50) into Equation (52),  $\dot{V}$  can be expressed as [Equation (53)]

$$\dot{V} = e_x^t \cdot g(x, u) \cdot e_d - e_x^t \cdot \mathbf{S} \cdot e_x$$
  
+ $e_d^t \cdot \mathbf{K}_p \cdot g(x, u) \cdot e_d - e_d^t \cdot \mathbf{K}_p \cdot \mathbf{S} \cdot e_x$   
+ $e_d^t \cdot \mathbf{K}_i \cdot e_x - e_d^t \cdot g^t(x, u) \cdot e_x$  (53)

Consequently, By referring to Equations (47) and (48), it yields

$$\dot{V} = \frac{1}{2} \cdot \begin{pmatrix} e_x & e_d \end{pmatrix} \cdot \begin{pmatrix} -\mathbf{S} & 0 \\ 0 & -\mathbf{P} \end{pmatrix} \cdot \begin{pmatrix} e_x \\ e_d \end{pmatrix}$$
(54)



Figure 6: Test bench setup PMSM's drive system.

From Equations (51) and (54), the exponentially stability of the estimation is satisfied long as constant matrix S and P are positive-definite matrix.

Finally, the constant matrix **S** and **P** have to be tuned by the assumption that the dynamics response of the state error  $e_x$  is highly faster than the error of the unknown parameters  $e_d$ . Consequently, the constant matrix **S** with eigenvalues real parts is greater than **P**. The experimental tuning is carried out to assure that the convergence of the state error to zero.

#### 3 Simulation and Experimental Validate

#### 3.1 Laboratory setup

The main PMSM parameters are presented in Table 1, and the flatness controller parameters are defined in Table 2. The laboratory setup showing in Figure 6 composed of a 6-pole, 1-kW PMSM coupled with a 0.25-kW Separate Excited DC motor that was served as a power supply for a purely resistive load. The stator windings of the PMSM were fed by a 3-kW,  $3\Phi$  DC-AC voltage-source inverter (VSI) that was operated at a switching frequency of 10 kHz. The input voltage is obtained through diode rectifier as shown in Figure 1. The drive system was also equipped with an incremental encoder mounted on the rotor shaft and has a resolution of 4096 lines/ revolution. The measurement of the electromagnetic torque was realized by a torque transducer mounted on the rotor shaft.

Symbol	Meaning	Value
Trated	Torque Rated	3 Nm
р	Number of Poles pair	3
$R_s$	Resistance (Motor + Inverter)	10.1 Ω
$L=L_d=L_q$	Stator inductance	35.31 mH
$\psi_m$	Magnetic flux	0.2214 Wb
J	Equivalent inertia	0.0022 kg.m <sup>2</sup>
В	Viscous friction coefficient	$3.5 \times 10^{-3}$ Nm.s/rad
$v_{BUS}$	DC Bus voltage	530 V
$f_s$	Switching frequency	10×10 <sup>3</sup> Hz

Table 1: PMSM/Inverter and parameters

 Table 2: Control parameters

Heading 1	Heading 2	Heading 3
$\zeta_1$	Damping ratio 1	1 pu.
$\omega_{n1}$	Natural frequency 1	3200 Rad.s <sup>-1</sup>
$\zeta_2$	Damping ratio 2	1 pu.
$\omega_{n2}$	Natural frequency 2	320 Rad.s <sup>-1</sup>
$\zeta_3$	Damping ratio 3	0.7 pu.
ω <sub>n3</sub>	Natural frequency 3	32 Rad.s <sup>-1</sup>
$\zeta_4$	Damping ratio 4	0.7 pu.
<i>ω</i> <sub>n4</sub>	Natural frequency 4	32 Rad.s <sup>-1</sup>
i <sub>qmax</sub>	The max. quadrature current	+6 A
i <sub>qmin</sub>	The min. quadrature current	-6 A

## 3.2 Performance of speed acceleration

Figure 7 depicts the experimental results of speed acceleration response at light load condition (friction losses),  $n_{COM}$  0–1500 rpm and  $i_{dCOM} = 0$  A. It is important to mention that the motor speed is able to track approximately 100% the command. During the acceleration period, the *q*-axis  $i_q$  equals the motor maximum capability ( $i_{qmax}$  = +6 A). This ensures that the PMSM runs up in the shortest time possible, and subsequently, the current  $i_q$  decreases in order to satisfy the small friction torque.

## 3.3 Performance of state variables estimation

To validate that the controller is appropriately designed, a simulation of the controller was developed using MATLAB/SIMULINK software. Figure 8(a) and (b) show the simulation and experimental results respectively of  $T_L$  estimation by using extended Luenberger observer. In Figure 8(a) and (b), Ch1 is the



Figure 7: Experimental result of speed acceleration.

torque reference  $T_{LREF}$ , Ch2 is the measured speed n, Ch3 is the q-axis current  $i_q$ , and Ch4 is the estimated torque  $T_{Lest}$ . The simulation and preliminary results indicate that both are corresponding. The results reflect that when the external disturbance torque is suddenly set from 0 nm to 2 nm, it can be correctly estimated by ELO, and the converging time is less than 0.1 s.

Figure 9(a) shows the simulation results and Figure 9(b) shows the experimental results, respectively of external torque load by using the new observer. In Figure 9(a) and (b), Ch1 is the torque reference  $T_{LREF}$ , Ch2 is the measured speed *n*, Ch3 is the *q*-axis current



**Figure 8**: Simulation and experimental results: External torque load estimation response by using ELO.

 $i_q$ , and Ch4 is the estimated torque  $T_{Lest}$ . The results reflect that when the external disturbance torque is suddenly taken from 0 nm to 2 nm, it can be correctly estimated by the exponentially stable, and the converging time is less than 0.04 s that is distinctly better than ELO.

#### 3.4 Speed reversal of flatness-based controller

The experimental results of speed reversal responses of the system are illustrated in Figure 10, where the motor is forced to reverse its direction.



**Figure 9**: Simulation and experimental results: External torque load estimation by using the new observer.

In Figure 10(a), Ch1 is the speed command  $n_{COM}$ , Ch2 is the measured speed *n*, Ch3 is *q*-axis current  $i_q$ , and Ch4 is the speed reference  $n_{REF}$ . In Figure 10(b), Ch5 is *d*-axis current  $i_d$ , Ch6 is the phase A current  $i_A$ , and Ch7 is the phase C current  $i_C$ . The system operates in a regenerative mode until the speed of the rotor will become positive; and thereafter, the system changes to motoring mode until the rotor speed reaches reference value. The experimental results reflect that the speed of PMSM can efficiently be controlled by flatnessbased control. During steady-state region, the speed measurement is able to almost 100% track the speed





Figure 10: Experimental results of speed reversal.

reference and the speed command, and q-axis current is restrained without exceeding the current limitation (+6 Ampere).

# 3.5 Performance of disturbance rejection

In Figure 11(a), Ch1 is  $v_{iq}$ , Ch2 is the measured speed n, Ch3 is the q-axis iq, and Ch4 is the estimated external torque  $T_{Lest}$ . In Figure 11(b), Ch5 is d-axis current  $i_d$ , Ch6 is the phase A current  $i_A$ , Ch7 is the phase C current  $i_C$ . Figure 11(c) shows the trajectories of the transient stator current. To guarantee the stability of the proposed control scheme, the responses of the speed controller and current controller as well as the disturbance rejection ability have been depicted in this section. The new



Figure 11: Experimental results of disturbance rejection.

observer was chosen for estimating the external torque disturbance and the losses of the control system.

The results indicate that when toque command is set to 2 Nm it is able to be satisfyingly estimated by the new observer and  $v_{tq}$  is completely observed as well. The reversing time of the speed is lower than 0.1 sec as well as *q*-axis has a rapid response.

## 4 Conclusions

This paper presents modeling of permanent magnet motor with parameters identify based on extended Luenberger observer and a new observer method. To control the permanent magnet synchronous motor for high-performance operation, it is still challenging due to its nonlinear properties and unknown parameters. Therefore, to deal with this issue, nonlinear control with parametric identification can offer high-performance control.

Both simulations by MATLAB/SIMULINK software and the laboratory experiments were carried out in this research. A small-scale PMSM was used to validate the proposed control scheme. The simulation and experimental results reflect that flatness-based control can offer the high-performance control of PMSM. Indeed, it provided a satisfying response to both current and speed control. Furthermore, parameters and state variables of the control system were correctly estimated by proposed observer. Moreover, the proposed control scheme can easily be carried out even if with other machines such as induction motor, brushless DC motor, etc.

## Acknowledgments

This work was supported in part by the research program in cooperation with the Faculty of Technical Education and the Thai-French Innovation Institute, King Mongkut's University of Technology North Bangkok, with the Universit'e de Lorraine, under Grant KMUTNB-61-GOV-D-68.

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